

# CONVECTIVE HEAT TRANSFER FROM/ TO FLUID FLOWING IN A STRAIGHT CIRCULAR DUCT ( $Re < 2000$ )

- Quantity of interest:
  - Duct length required to raise mean fluid temperature from  $T_0$  to  $T_b < T_w$
  - $U \rightarrow$  average axial fluid velocity
  - $\rho_0(\pi d_w^2/4)U \rightarrow$  total mass flow rate
  - $v_z(r)$  parabolic (constant property flow Newtonian fluid far downstream of inlet)
  - $T(r)$  then satisfies linear, 2<sup>nd</sup>-order PDE:

$$2U \cdot \left[ 1 - \left( \frac{2r}{d_w} \right)^2 \right] \frac{\partial T}{\partial z} = \frac{\alpha}{r} \cdot \frac{\partial}{\partial r} \left( r \frac{\partial T}{\partial r} \right)$$

(axial conduction neglected compared with radial)

# CONVECTIVE HEAT TRANSFER FROM/ TO FLUID FLOWING IN A STRAIGHT CIRCULAR DUCT ( $Re < 2000$ )

- Boundary conditions:

$$T(r, 0) \equiv T_0 = \text{constant}$$

$$T(d_w / 2, z) \equiv T_w = \text{constant}$$

$$\left( \frac{\partial T}{\partial x} \right)_{r=0} = 0,$$

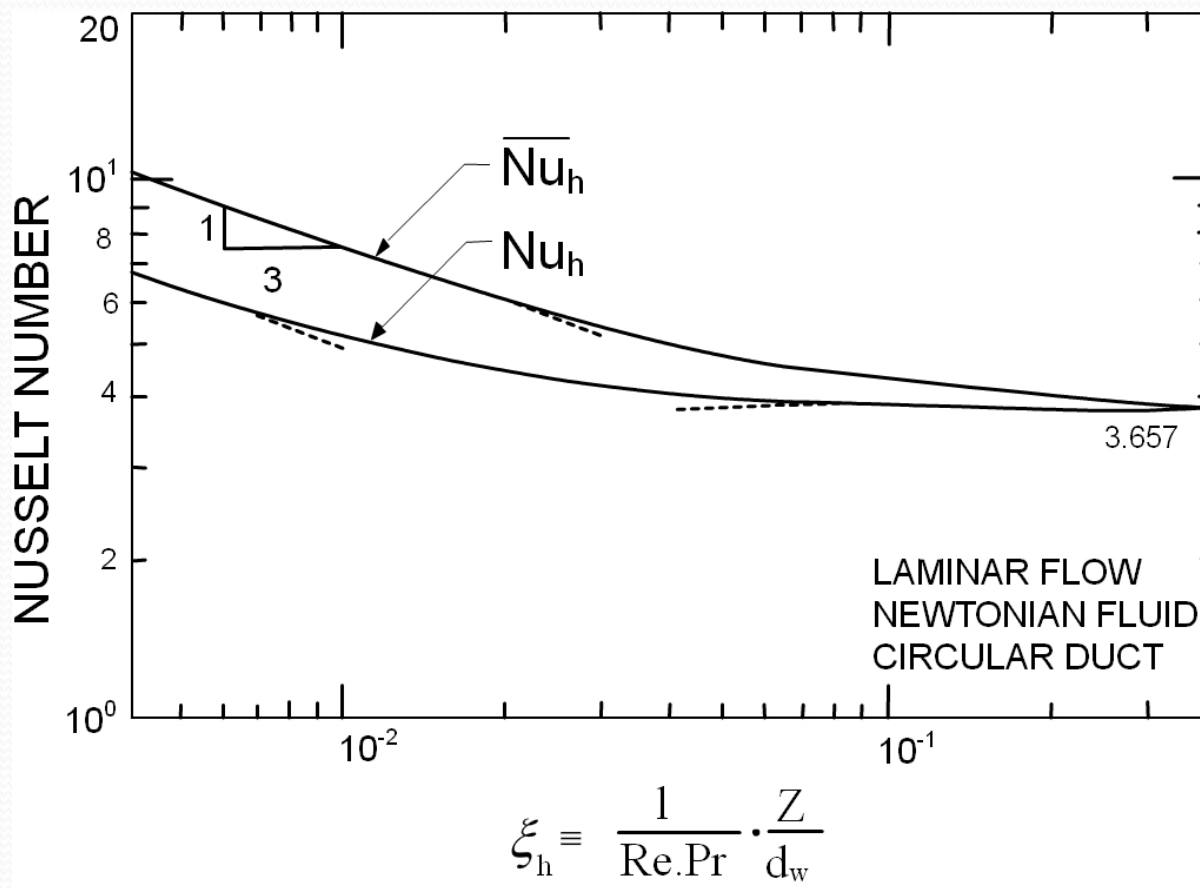
- $T(r, z)$  found by Fourier method of “Separation-of-variables”
- Local heat flux

$$\dot{q}_{ref}'' \equiv k \left[ \frac{T_w - T_b(z)}{d_w} \right]$$

# CONVECTIVE HEAT TRANSFER FROM/ TO FLUID FLOWING IN A STRAIGHT CIRCULAR DUCT ( $Re < 2000$ )

- $T_b(z)$  → mixing-cup average temperature at axial station  $z$   
Fluid temperature that would result from severing the duct at  $z$  and adiabatically mixing the effluent

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# CONVECTIVE HEAT TRANSFER FROM/ TO FLUID FLOWING IN A STRAIGHT CIRCULAR DUCT ( $Re < 2000$ )

- $Nu_h$  is a function of the dimensionless axial variable  
(Graetz, 1885)

$$\xi_h \equiv \frac{1}{Re Pr} \cdot \frac{z}{d_w}$$

and

$$Nu_h \approx const \xi_h^{-1/3} \quad \text{for } \xi_h \rightarrow 0$$

$$Nu_h \rightarrow 3.657 \quad \text{for } \xi_h \rightarrow \infty$$

# CONVECTIVE HEAT TRANSFER FROM/ TO FLUID FLOWING IN A STRAIGHT CIRCULAR DUCT ( $Re < 2000$ )

- $Nu_h \rightarrow \infty$  at thermal inlet
- $Nu_h \rightarrow$  constant value far downstream (fully-developed or asymptotic Nusselt number)

$$\overline{Nu}_h(\xi_h) \equiv \frac{1}{\xi_h} \cdot \int_0^{\xi_h} Nu_h(\xi_h') d\xi_h'$$

Axial average heat-transfer coefficient

$$\overline{Nu}_h(\xi_h) = \overline{Nu}_h(\infty) \cdot \left[ F \left( \begin{array}{c} \text{thermal} \\ \text{entrance} \end{array} \right) \right]$$

where

$$F(\text{thermal entrance}) \approx \left[ 1 + (7.60\xi_h)^{-8/3} \right]^{1/8}$$

# CONVECTIVE HEAT TRANSFER FROM/ TO FLUID FLOWING IN A STRAIGHT CIRCULAR DUCT ( $Re < 2000$ )

- Overall energy balance on fluid contained in Eulerian CV between 0 and z yields:

$$\frac{T_w - T_b(z)}{T_w - T_b(0)} = \exp\left\{-4\xi_h \overline{Nu}_h(\infty) \cdot F(\text{thermal entrance})\right\}$$

- Enables calculation of length required to achieve any  $T_b$

# CONVECTIVE HEAT TRANSFER FROM/ TO FLUID FLOWING IN A STRAIGHT CIRCULAR DUCT ( $Re < 2000$ )

Alternatively:

$$\frac{\dot{q} / A_w}{\left\{ \frac{k (\Delta T)_{mean}}{d_w} \right\}} = \overline{Nu}_h (\xi_{max})$$

where

$$\dot{q} = \dot{m} c_p [T_b(L) - T_b(0)]$$

and

$$A_w = \pi d_w L$$



# CONVECTIVE HEAT TRANSFER FROM/ TO FLUID FLOWING IN A STRAIGHT CIRCULAR DUCT ( $Re < 2000$ )

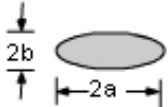
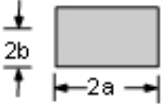
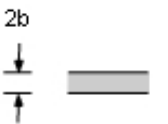
- Resulting in:

$$\Delta T_{mean} \equiv \frac{(T_w - T_b(0)) - (T_w - T_b(L))}{\ln \left[ \frac{T_w - T_b(0)}{T_w - T_b(L)} \right]} \equiv LMTD$$

(log-mean temperature difference, LMTD)

- Other important internal-flow heat-transfer problems exhibit same general features:





# CONVECTIVE HEAT TRANSFER FROM/ TO FLUID FLOWING IN A STRAIGHT CIRCULAR DUCT ( $Re < 2000$ )

Geometry	b/a	$Nu_T$	$Nu_{H1}$	$Nu_{H2}$	$C_f Re$	$\frac{j_{H1}^+}{C_f}$	$\frac{Nu_{H1}}{Nu_T}$
	1	3.657	4.364	4.364	16.000	0.307	1.19
	1/2	3.74	4.88		18.24	0.301	1.30
	1/8	3.72	5.09		19.15	0.299	1.37
	1	2.976	3.608	3.091	14.227	0.286	1.21
	1/2	3.391	4.123	3.017	15.548	0.299	1.22
	1/4	4.439	5.331	2.930	18.233	0.329	1.20
	1/8	5.597	6.490	2.904	20.585	0.355	1.16
	0	7.541	8.235	8.235	24.000	0.386	1.09

† This heading is the same as  $Nu_{H1} Pr^{1/3} / C_f Re$  with  $Pr = 0.7$ ;  $Nu, C_f, Re$  based on  $d_{eff} \equiv 4A / \text{perimeter}$   
a The subscripts T, H1 and H2 on the relevant Nusselt numbers mean, respectively:

- T1: pertaining to constant wall temperature
- H1: pertaining to actual constant heat flux
- H2: pertaining to peripheral and axial constancy of heat flux

# CONVECTIVE HEAT TRANSFER FROM/ TO FLUID FLOWING IN A STRAIGHT CIRCULAR DUCT ( $Re < 2000$ )

Geometry	$b/a$	$Nu_T$	$Nu_{H1}$	$Nu_{H2}$	$C_f Re$	$\frac{j_{H1}^+}{C_f}$	$\frac{Nu_{H1}}{Nu_T}$
 insulated	0	4.861	5.385	5.385	24.000	0.253	1.11
		3.34*	4.002	3.862	15.054	0.299	1.20
	$\sqrt{3}/2$	2.39*	3.014	1.474	12.630	0.269	1.26
	$\sqrt{3}/2$	2.47	3.111	1.892	13.333	0.263	1.26

† This heading is the same as  $Nu_{H1} Pr^{-1/3} / C_f Re$  with  $Pr = 0.7$ ;  $Nu, C_f, Re$  based on  $d_{eff} \equiv 4A / \text{perimeter}$   
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# HEAT EXCHANGE BETWEEN A FLUID AND A POROUS MEDIUM (“PACKING”)

- Packed duct is equivalent of many tortuous interstitial ducts defined by spaces between bed particles
- Appropriate dimensionless energy-transport coefficient →

$Nu_{h,bed}$

- Depends on  $Re_{bed}$ ,  $Pr$

$$Re_{bed} \equiv \frac{G_o d_{p,eff}}{\mu (1 - \varepsilon)}$$

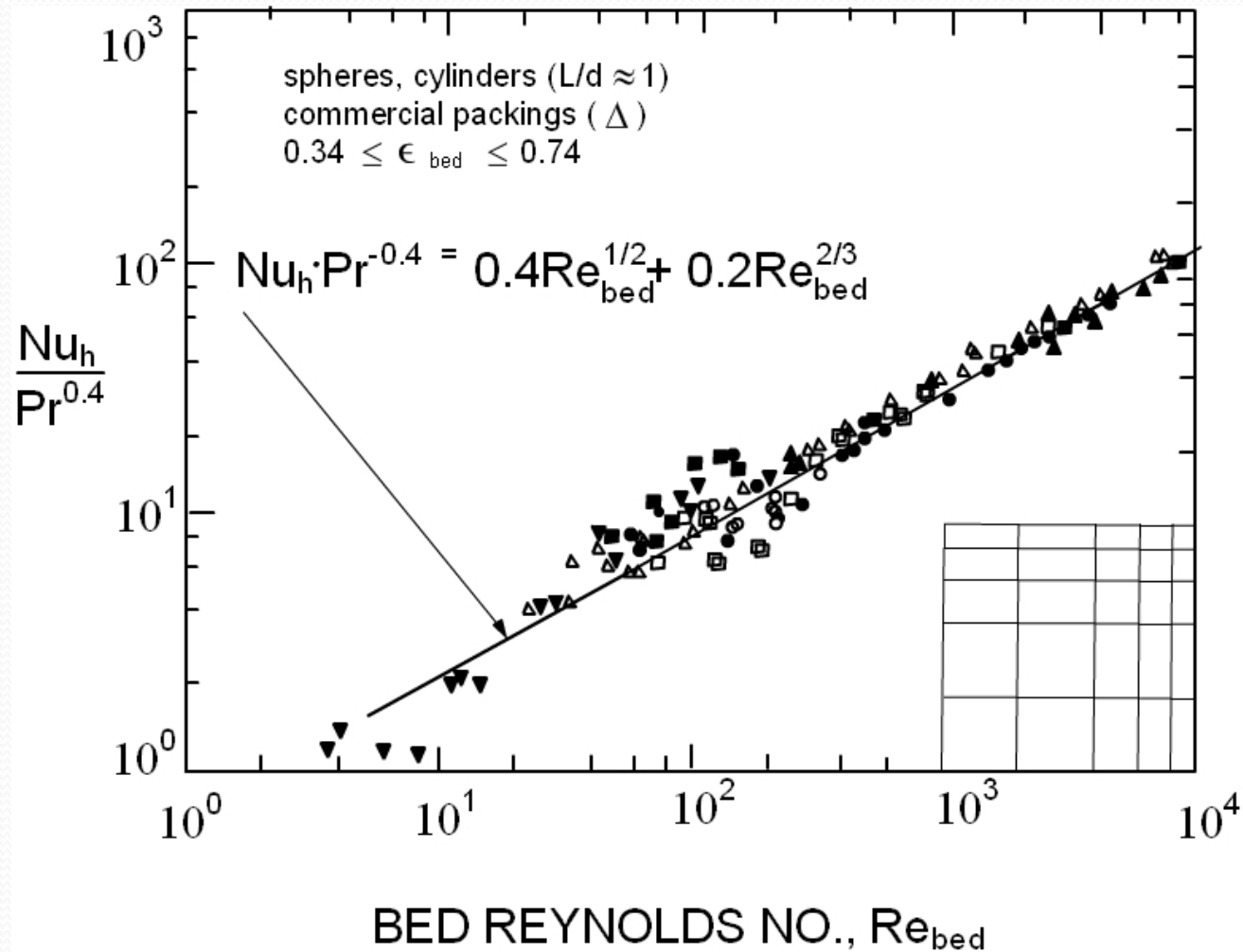
$G_o (\equiv \dot{m} / A_o) \rightarrow$  “superficial” fluid mass velocity

# HEAT EXCHANGE BETWEEN A FLUID AND A POROUS MEDIUM (“PACKING”)

- $dz \rightarrow$  element of column height
- Total volume  $\rightarrow A_0 dz$
- Packing surface area  $\rightarrow a''' A_0 dz$
- Energy transferred from packing to fluid  $\rightarrow d\dot{q}$
- Average heat flux  $\rightarrow d\dot{q} / (a''' A_0 dz)$
- Dimensionless heat transfer coefficient

$$Nu_{h,bed} = const. \frac{d\dot{q} / (a''' A_0 dz)}{\left[ k (T_w - T_b) \right] / d_{i,eff}}$$

# HEAT EXCHANGE BETWEEN A FLUID AND A POROUS MEDIUM (“PACKING”)



## HEAT EXCHANGE BETWEEN A FLUID AND A POROUS MEDIUM (“PACKING”)

- Analogous expressions apply for species mass exchange between fluid & fixed/ fluidized bed, tube bundle
  - e.g., fixed-bed catalytic reactor
  - Preliminary design based on “plug flow” in cylindrical vessel of constant cross-section
  - Refinements needed to take care of non-idealities
    - Radially non-uniform  $G_0$ , non-uniform packing temperature, variable fluid properties, etc.

# DIFFUSION FROM A STEADY POINT SOURCE IN A UNIFORM MOVING STREAM

- Quantity of interest:
  - Thermal wake of point heat source of strength  $\dot{q}$ 
    - In constant-property, uniform, laminar stream
- Relevant to combustion gases, distributed heat sources, pollutant transport (mass-transfer analog)
- Energy is convected downstream (+z direction), diffuses radially & axially



# DIFFUSION FROM A STEADY POINT SOURCE IN A UNIFORM MOVING STREAM

- Local energy-balance equation:

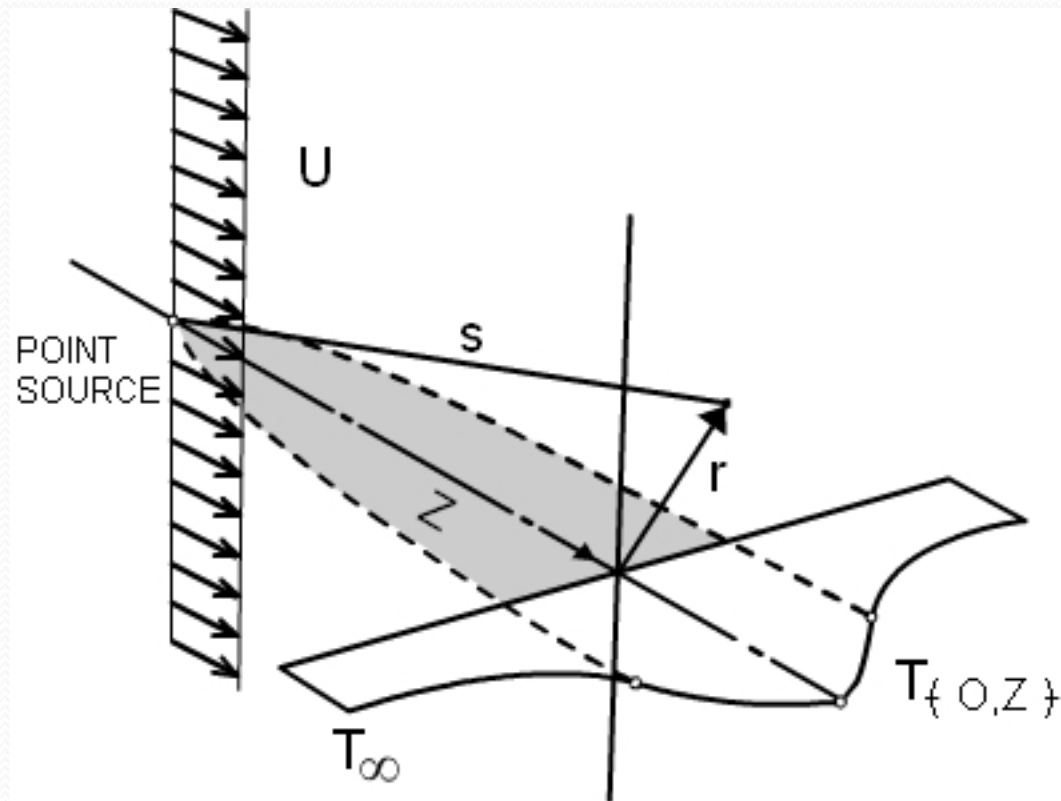
$$U \frac{\partial T}{\partial z} = \alpha \left[ \frac{1}{r} \frac{\partial}{\partial r} \left( r \frac{\partial T}{\partial r} \right) + \frac{\partial^2 T}{\partial z^2} \right] \quad (\text{if } \alpha = \text{const}).$$

Spherical radius

$$s \equiv (r^2 + z^2)^{1/2}$$

(measured from point source)

# DIFFUSION FROM A STEADY POINT SOURCE IN A UNIFORM MOVING STREAM



Energy convection and diffusion from a continuous point source in a uniform stream

# DIFFUSION FROM A STEADY POINT SOURCE IN A UNIFORM MOVING STREAM

➤ Boundary conditions:

$$\lim_{s \rightarrow \infty} T = T_{\infty} \quad (\text{specified constant})$$

$$\lim_{s \rightarrow 0} \left[ - (4\pi s^2) k \left( \frac{\partial T}{\partial s} \right) \right] = \dot{q} \quad (\text{specified source strength})$$

$$\lim_{r \rightarrow 0} \left( \frac{\partial T}{\partial r} \right) = 0 \quad \text{for } z > 0$$

Leading to:

$$T(r, z) - T_{\infty} = \frac{\dot{q}}{4\pi ks} \cdot \exp \left[ \frac{-U(s - z)}{2\alpha} \right]$$

# DIFFUSION FROM A STEADY POINT SOURCE IN A UNIFORM MOVING STREAM

- Relation implies non-physical behavior:  $T \rightarrow \infty$  in immediate vicinity of point source

- Meaningful at downstream distances  $\gg$  source size

- At any  $z \gg r$ , radial temperature profiles are Gaussian

in shape: 
$$\frac{T(r, z) - T_\infty}{T(0, z) - T_\infty} \approx \exp \left[ -\frac{Uz}{4\alpha} \left( \frac{r}{z} \right)^2 \right]$$

- Along the axis  $r = 0$ ,  $T(0, z) - T_\infty = \frac{\dot{q}}{4\pi k z}$

- Can be used to determine k-values based on axial temperature decay data

- No htc's since there are no boundary surfaces



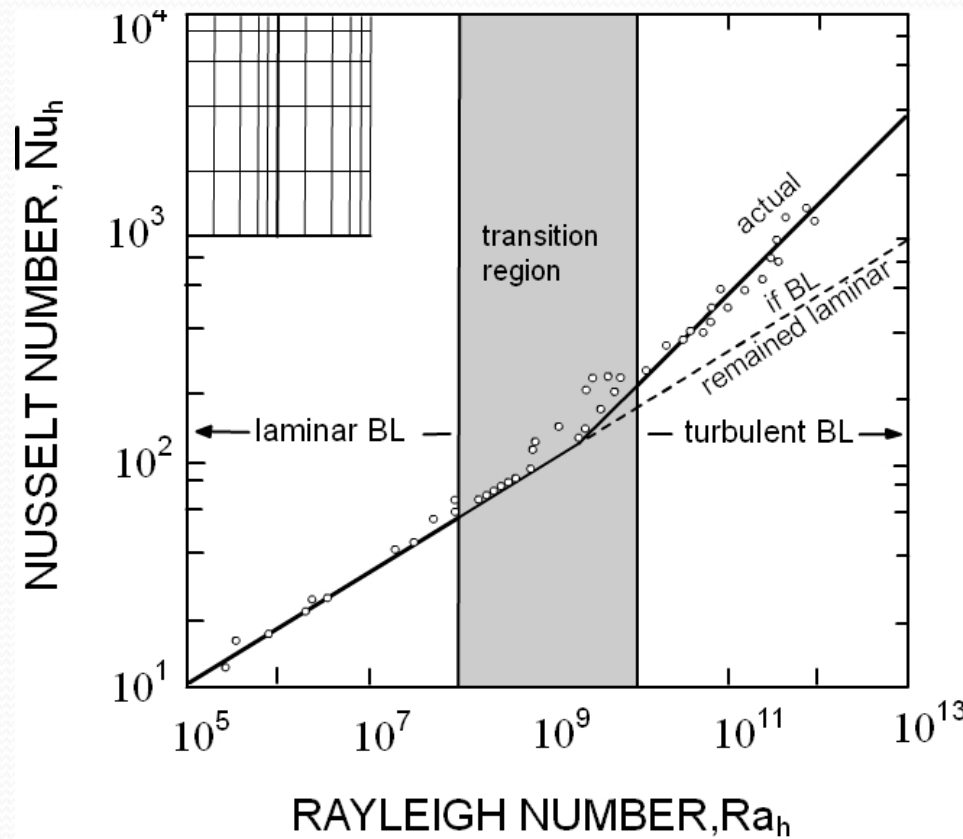
## STEADY TURBULENT FLOWS

- Turbulence: localized non-steadiness associated with enhancement of time-averaged rates of momentum, energy & mass transport
  - For  $Re$  and  $Ra_h$  numbers above “transition” values
  - Extrapolation of laminar heat-transfer coefficient laws into turbulent region will result in serious underestimation of transport rates

## STEADY TURBULENT FLOWS

- Turbulence: localized non-steadiness associated with enhancement of time-averaged rates of momentum, energy & mass transport
  - e.g., higher drag (undesirable), higher heat-exchange rates (desirable)– the two may be in conflict!
  - For  $Re$  and  $Ra_h$  numbers above “transition” values
  - Extrapolation of laminar heat-transfer coefficient laws into turbulent region will result in serious underestimation of transport rates

# STEADY TURBULENT FLOWS



Vertical flat surfaces in quiescent Newtonian fluid

# FORCED-CONVECTIVE TURBULENT HEAT TRANSFER FROM/ TO STRAIGHT, SMOOTH DUCTS

- Fully-developed Nusselt numbers not constant, but depend on Re, Pr

$$Nu_h(\infty) = \overline{Nu_h}(\infty) \cong 0.023 Re^{0.8} Pr^{1/3}$$

or

$$St_h(\infty) = \overline{St_h}(\infty) \cong 0.023 Re^{-0.2} Pr^{-2/3}$$

(for  $Re > 10^4$ ,  $Pr \geq 0.7$ )

- Apply to non-circular ducts as well (with  $d_{\text{eff}} = 4A/P$ )



# FORCED-CONVECTIVE TURBULENT HEAT TRANSFER FROM/ TO STRAIGHT, SMOOTH DUCTS

- Thermal entrance effect is modified to:

$$F(\text{thermal entrance}) \approx 1 + \left( \frac{z}{d_w} \right)^{-2/3} \quad (\text{Re} > 10^4, \text{Pr} > 0.6)$$

- Above Re, Pr-dependencies are oversimplifications
- Stanton number and pipe friction factor are closely related (extended Reynolds' analogy)

# TURBULENT THERMAL BOUNDARY LAYERS ON A SMOOTH FLAT PLATE

- For  $Re_L > 10^5$ , viscous flow within a flat-plate forced-convection BL becomes turbulent at some  $x$  (transition)
  - Heat transfer across laminar BL for  $x < x(\text{transition})$
  - Heat transfer across turbulent BL for  $x(\text{transition}) \leq x \leq L$
- If turbulence is triggered near leading-edge, turbulence dominant over entire plate, and for  $10^5 \leq Re_L \leq 10^7$ , skin friction distribution becomes:

$$\frac{\tau_w(x)}{\rho_\infty U^2} \equiv \frac{1}{2} c_f(x) = 0.0288 (Re_x)^{-1/5}$$

# TURBULENT THERMAL BOUNDARY LAYERS ON A SMOOTH FLAT PLATE

➤ Stanton-number distribution:

$$St_h(x) \cong 0.0288(\text{Re}_x)^{-1/5} \text{Pr}^{-2/3}$$

or

$$\overline{Nu}_h(x) \cong 0.0288(\text{Re}_x)^{4/5} \text{Pr}^{1/3}$$

Averaging over entire plate on one side:

$$\overline{Nu}_h \cong 0.036(\text{Re}_L)^{4/5} \text{Pr}^{1/3}$$

# TURBULENT THERMAL BOUNDARY LAYERS ON A SMOOTH FLAT PLATE

- Thermal turbulent BL slope thickness  $\delta_h \cong x^{1/5}$ 
  - Laminar:  $\delta_h \cong x^{1/2}$
- Time-averaged heat fluxes in turbulent BL fall off with distance as  $x^{-1/5}$
- On non-isothermal curved surfaces,  $x \rightarrow$  distance along surface
  - e.g., turbine blades, nozzle surfaces

# TURBULENT THERMAL BOUNDARY LAYERS ON A SMOOTH FLAT PLATE

- Natural convection:

- For Newtonian fluids with Prandtl numbers nearly equal to that of air, and  $Ra_h > 10^9$ :

$$Nu_h(x) \cong 0.10(Ra_h)^{1/3} \quad (\text{Pr} \approx 0.7)$$

- Local heat fluxes nearly constant along plate surface, hence:

$$\overline{Nu_h} \cong 0.10(Ra_h)^{1/3} \quad (\text{for } \text{Pr} \approx 0.7, \quad Ra_{h,L} > 10^9)$$

- (reference length  $\rightarrow L$ )

## ISOLATED SPHERE AT HIGH Re

- For  $Re \gg 3 \times 10^5$ , turbulence within thermal BL amplifies total heat-transfer coefficient,  $Nu_h$ , for an isothermal sphere

- Up to  $Re = 10^5$  :

$$\overline{Nu_h} = 2 + \left[ 0.4(Re)^{1/2} + 0.06(Re)^{2/3} \right] (Pr)^{0.4}$$

- For  $Re \gg 3 \times 10^5$ , reliable data & correlations not available
- Separated flow introduces great complexity in theory



## ISOLATED SPHERE AT HIGH $Re$

- Turbulence types:
  - Confined to BL adjacent to solid surfaces
  - Mainstream turbulence
- Empirical correction factors needed for:

$I_{t,\infty}$  = mainstream turbulence “intensity” (ratio of root-mean-square (rms) velocity fluctuation to time-averaged approach velocity),



## ISOLATED SPHERE AT HIGH $Re$

and

$L_{t,\infty} / L \equiv$  ratio of mainstream turbulence (macro-) scale  
to body dimension

- Mainstream turbulence has two effects:
  - Triggers earlier transition to turbulence within BL
  - Modifies time-averaged transport across laminar BL